

SUPPLEMENTAL TYPE CERTIFICATE

10055068

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to:

WHELEN ENGINEERING COMPANY Inc.

51 WINTHROP ROAD
CHESTER CT 06412
USA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: SEE EASA APPROVED MODEL LIST

Type Certificate Holder: SEE EASA APPROVED MODEL LIST

Type: SEE EASA APPROVED MODEL LIST

Model: SEE EASA APPROVED MODEL LIST

Original STC Number: FAA STC SA800EA

Description of Design Change:

EASA validation of FAA STC SA800EA: Installation of Whelen Anti-Collision, Position, or Antidesign change (clear) Collision/Position Light System- limited to EASA GA aeroplanes (Approved Model List)

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Date of Issue: 09 October 2015


Yves MORIER

**Head of General Aviation and
Remotely Piloted Aircraft Systems (RPAS)**

10039344

SUPPLEMENTAL TYPE CERTIFICATE - 10055068 - WHELEN ENGINEERING COMPANY Inc. - 306507



An Agency of the European Union

Associated Technical Documentation:

Whelen Anti-Collision Light Systems Installation and Service Manual, Document No. 05131, Rev. D dated May 2015 or later approved revision
or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/ or the Technical Implementation Procedures of EU/ USA Bilateral Agreement.

Limitations/Conditions:

Limited to EASA Approved Model List (attached)

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -